MANDATORY INFORMATION FOR UPLOADING OF RFI/RFP

Ser	Title	Information
1.	Tender Title	Request for information(RFI) for setting of Naval Aircraft and Ship
		Instrumentation and Telemetry(NASI)
2.	Tender Reference Number	AO/10003/NASI
3.	Tender Type	Buy
4.	Bid Process	Two Bids
5.	Location	New Delhi
6.	First Announcement	27 Jun 18
7.	Last Date of Document Collection	NA
8.	Last Date of Submission	31 Aug 18
9.	Opening Date	NA
10.	Brief Work Description	IN intends to setup Naval Aircraft and Ship Instrumentation and Telemetry (NASI) Equipment. Instrumentation and telemetry hardware and software setup is an integral part of flight trials and is required for quantitative analysis. The IN requires ship and shore based compatible NASI setup.
11.	Contract address	The Principal Director Naval Air Staff Integrated Headquarters of Ministry of Defence(Navy) Directorate of Naval Air Staff(DNAS) Room No.85,'A' Block Hutments Dalhousie Road New Delhi 110011,India
12	Contact Telephone	011-23010592/ 011-23010940
13.	Date of Vendor Interaction	1000 hrs on 27 Jul 18

Tele: +91-11-23011008 Fax: +91-11-23010351

> Directorate of Naval Air Staff Integrated Headquarters Ministry of Defence (Navy) 'A' Block Hutments New Delhi - 1100011

AO/10003/NASI

Jun 18

REQUEST FOR INFORMATION (RFI) FOR SETTING OF NAVAL AIRCRAFT AND SHIP INSTRUMENTATION AND TELEMETRY (NASI) EQUIPMENT FOR THE INDIAN NAVY

1. The Ministry of Defence, Government of India, intends to setup Naval Aircraft and Ship Instrumentation and Telemetry (NASI) EQUIPMENT. *Instrumentation and telemetry hardware and software* setup is an integral part of flight trials and is required for quantitative analysis. Trial agencies such as ASTE/ IAF, NFTC/ ADA, CABS have similar setups. However, these setups are primarily oriented towards shore based testing. The *IN* requires to have a ship and shore based compatible NASI setup.

2. **<u>RFI Structure</u>**. This Request for Information (RFI) consists of two parts as indicated below: -

(a) **Part I**. This part of RFI incorporates the intended use of NASI and its features that should be met by the vendor(s). Vendor specifications and other requirements as per Defence Procurement Procedure (DPP) 2016 are discussed in this part.

(b) **Part II**. The second part of the RFI states the methodology of seeking response of vendors / vendors. Submission of incomplete response format will render the vendors / vendors liable for rejection.

<u> Part - I</u>

3. **Explanation of Terms**. Following terminology would be used in the documents: -

(a) <u>Home Location (HL)</u>. Naval Flight Test Squadron (NFTS), INAS 552, Goa would be the home location (HL) for all trials. This is the location where all resources are required to be created and primarily utilised.

(b) <u>**Trial Location (TL)**</u>. Places where the trials are planned to be conducted are the trial location (TL). The TL could be any Naval, Army, Air Force, DRDO base, location of a defence PSU or as designated by the Indian Navy. Ships of the Indian Navy could also be nominated as TL.

4. **Ship Instrumentation and Telemetry (NASI)**. NASI facilitates generation of data during ground and flight trial. NASI consists of instrumentation and telemetry setup on ground, ship and aircraft. With data gathered by NASI, quantitative assessments are performed.

5. <u>Composition of NASI</u>. NASI consists of following five modules:-

Fixed Wing and Carrier Suitability Trial Module. This will (a) module cover all fixed wing trials as well trials related to aircraft carrier compatibility. This module will cater for fighter as well as MR aircraft towards their ground and flight trials. During carrier compatibility trial of fight aircraft, high sink rate landings, ski jumps and arrested recoveries under various configurations would be performed which necessitate instrumentation of ship's equipments such as arresting gear, restraining gear, deck for motion and navigational parameters etc. Further, the aircraft is required to be instrumented for generation of stability, performance parameters. Details of parameters to be captured are placed at Appendix 'A' and 'B'. Accordingly, aircraft carrier equipment and one fixed wing aircraft are required to be instrumented and have telemetric capability. Details for utilization of this module is enumerated below:-

(i) <u>Platforms for Instrumentation and Telemetry</u>. Instrumentation and telemetry with real time data storage and indication needs to be incorporated on following platforms with capability to display own as well as telemetered data:-

(aa) <u>Aircraft</u>. Instrumentation consisting of integration of sensors, data acquisition system, displays with customised

monitoring for parameters, event marking provision needs to be incorporated on a nominated fixed wing aircraft. List of parameters which are required to be monitored is placed at Appendix 'A'. Each of these parameters are required to be telemetered to the ship. Parameters telemetered from the ship are required to be cross displayed on the aircraft in user configurable format.

(ab) <u>Aircraft Carrier</u>. Instrumentation of equipment of aircraft carrier consisting of integration of sensors, data acquisition system, displays with customised monitoring for parameters, event marking provision needs to carried out. These equipment would be required to be portable for usage on either of aircraft carriers as well as usage as ground station. The telemetry equipment are required to relay information on real time basis to aircraft along with GPS information. List of parameters which are required to be monitored is placed at Appendix 'B'. Each of these parameters are required to be telemetered to the aircraft. Parameters telemetered from the aircraft are required to be cross displayed on the console in user configurable format.

(ii) <u>Types of Trials</u>. Instrumentation and Telemetry requirements for this module consist of following types of trials:-

<u>Ser</u>	Trial
(aa)	Stability Trial ¹
(ab)	Performance Trial ²
(ac)	Structural Integrity Evaluation ³
(ad)	Store Integration and Separation ⁴
(ae)	Avionics Evaluation ⁵
(af)	Parametric Identification Trials ⁶
(ag)	Miscellaneous Trials ⁷

¹ Stability trial is undertaken to ascertain that modifications to aircraft has not significantly altered stability of the aircraft.

² Performance evaluation is performed to bring out effect of modification on performance of aircraft w.r.t. Range, Endurance, Cruise, Take off, landing, hover (for helicopters), Climb and Descent.

³ Structural integrity assessment is to ascertain that modification has not caused adverse effect on structural soundness of the aircraft. This caters for limit load, flutter and fatigue testing.

⁴ Store integration trials are done to quantify safe adaptation, carriage, release, separation and accuracy of weapon from an aircraft.

⁵ Avionics trials cater for performance of avionics for the intended role of the system.

⁶ Parametric Identifications trials are done to determine aero-derivatives of the aircraft.

⁷ These trials involve usage of external camera pod for various trials like store release/ tyre trials etc.

(b) <u>Helicopter and SHOL Trials Module</u>. This module will cover all RW trials including Ship's Helicopter Operating Limits (SHOL) trials towards assessing the compatibility of helicopter with the ship. Details for utilisation of this module is enumerated below:-

(i) <u>Platforms for Instrumentation and Telemetry</u>. Instrumentation and telemetry with real time data storage and indication needs to be incorporated on following platforms:-

(aa) <u>Helicopter</u>. The scope of work involves integration of sensors, data acquisition system, telemetric equipment, displays (for monitoring for parameters) and event marker on the nominated helicopter. List of parameters which are required to be monitored is placed at Appendix 'C'. Each of these parameters are required to be telemetered to the ship as well as mobile telemetery. Parameters telemetered from the ship and mobile telemtry are required to be cross displayed on the aircraft in user configurable format.

(ab) <u>Mobile Telemetry Platform</u>. A mobile telemetry is required to be utilised towards real time data interpretation such as structural integrity, store separation, Pressure Error Correction (PEC) exercises and parametric identification trials. Many trials are undertaken at displaced locations owing to hot and cold weather trials, trials at EW range, armament ranges etc. Accordingly, there should be telemetry equipment installed on mobile platform for conduct of trials at displaced location (mobile platform should form part of the contract). The mobile telemetry should be capable of relaying parameters listed at Appendix 'D'. Each of these parameters are required to be telemetered to the ship as well as aircraft. Parameters telemetered from the ship and aircraft are required to be cross displayed on the mobile telemetry in user configurable format.

(ac) <u>Ship</u>. Instrumentation and telemetric equipment are required to be installed on ships to undertake SHOL trials. These equipment should be portable to be used on various ships. List of parameters which are required to be monitored is placed at Appendix 'E'. Each of these parameters are required to be telemetered to the aircraft as well as mobile telemetery. Parameters

telemetered from the aircraft and mobile telemtry are required to be cross displayed on the aircraft in user configurable format.

(ii) <u>Types of Trials</u>. Instrumentation and Telemetry requirements for this module consist of following types of trials:-

<u>Ser</u>	Trial
(aa)	SHOL Trials ⁸
(ab)	Stability Trial
(ac)	Performance Trial
(ad)	Structural Integrity Evaluation
(ae)	Store Integration and Separation
(af)	Avionics Evaluation
(ag)	Parametric Identification Trials
(ah)	Miscellaneous Trials

(c) **<u>Data Handling Module</u>**. Flight testing would generate immense amount of data. This module would be required to efficiently manage this data with prediction capability.

(d) <u>Aircraft Weighing Module</u>. Weighing of the aircraft is performed to determine the CG in all three (03) axes and weight. As handling qualities / performance are affected by weight and CG of the aircraft, weighing prior and post modification is a pre-requisite for handing qualities assessment. The vendor would be required to level a portion of 50 m patch at tarmac and provision minimum five (05) load cells to setup weighing provision at HL as per guidance by the Indian Navy.

(e) **Operation and Maintenance Module**. Operation, Maintenance and Warranty Support for all 4 modules above are required for duration of three (03) years w.e.f. supply of parts. The scope of work involves following:-

(i) Preparation of instrumentation scheme for ship and aircraft instrumentation.

- (ii) Obtaining approval from certification agency.
- (iii) Undertaking modification of aircraft / ship.

(iv) Configuring display console as per trial schedule and activating cognitive guidance.

(v) Download recorded data.

⁸ SHOL trial is undertaken to define operating envelope of a helicopter variant over ship.

(vi) Process downloaded data to engineering values.

(vii) Carryout data reduction for suitable for analysis under guidance of trial team.

- (viii) Undertake data analysis for prediction and generate report.
- (ix) Telemetered data must be displayed in specified format.

(x) The vendor should meet all requirements as brought out in preceding paragraphs. In addition, following conditions should be fulfilled by the vendor: -

(aa) The vendor should have a minimum 10 years of experience on Aircraft/Aerospace Instrumentation and telemetry.

(ab) Should have delivered Aviation Grade Instrumentation and telemetry products for last 20 years.

(ac) The vendor should be willing to share its balance sheet / yearly financial audit reports / turnover to ascertain its capacity to undertake the project.

(ad) The vendor should have a minimum strength of 50 engineers with relevant field experience in instrumentation / telemetry.

(ae) The vendor should be the OEM of Data Acquisition Unit.

<u>Part II</u>

6. **Procedure for Response**. The Vendor(s) fulfilling the criteria of Part I are required to respond to the parametric query as placed in Appendix 'F' along with their details as sought in Appendix 'G' and compliance matrix for Part I. Additional information on the product and product support facilities can be also furnished by the vendor(s).

7. <u>Vendor interaction</u>. Vendor interaction prior submission of responses will be held at 'A ' Block conference room, 'A' Block Hutments, Dara Sukoh Road,New Delhi-110011 on 27 Jul 18.

8 <u>Address for Response</u>. The filled form and the response (hard and soft copies) should be dispatched to the under mentioned address:-

The Principal Director Directorate of Naval Air Staff IHQ MoD (Navy) 'A' Block Hutments Dara Shukoh Road New Delhi 110011

Fax: 011-23010351

Contact Details: JDNAS 011- 23010592

9 Time for Response. Last date of receipt of response is <u>31 Aug 18</u>.

10 The Government of India invites responses to this RFI only from those vendors who meet the criteria mentioned at Para 5. Reply to this RFI (and further communication on the case, including equipment description, training and documentation) are to be made in English Language only. Response to the RFI is to be provided in hard and soft copy. The compliance tables to all aspects are required to be provided in editable form (preferably word format). The questionnaire is placed at **Appendix 'F'**.

11 This RFI is being issued with no financial commitment and the Ministry of Defence reserves the right to change or vary any part thereof at any stage. The Government of India also reserves the right to withdraw it, should it be so necessary at any stage.

<u>Appendix 'A'</u>

HARDWARE (SENSORS) REQUIRED FOR PARAMETERS TO BE CAPTURED USING INSTRUMENTATION AND TELEMETRY ON FIXED WING AIRCRAFT

<u>Ser</u>	<u>Data</u>	<u>Limits</u>	<u>Unit</u>	<u>Least</u> Count	<u>Sampling</u> <u>rate (Hz)</u>	<u>Stand</u> <u>Alone (S) /</u> Tapped (T)	<u>No of</u> <u>Sensors</u> per Aircraft
1.	Video	Up to 10 MP	MP	-	1000	Т	2
2.	Roll	-180-0-+180	Deg	0.1	1000	S	1
3.	Pitch	-90-0-+90	Deg	0.1	1000	S	1
4.	Yaw / Hdg	0-360	Deg	0.1	1000	S	1
5.	Linear acceleration along longitudinal axis (Nx)	3	G	0.1	1000	S	1
6.	Linear acceleration along lateral axis (Ny)	3	G	0.1	1000	S	1
7.	Linear acceleration along normal axis (Nz)	10	G	0.1	1000	S	1

<u>Ser</u>	<u>Data</u>	<u>Limits</u>	<u>Unit</u>	<u>Least</u> Count	<u>Sampling</u> rate (Hz)	<u>Stand</u> <u>Alone (S) /</u> <u>Tapped (T)</u>	<u>No of</u> <u>Sensors</u> per Aircraft
8.	Angular acceleration around longitudinal axis ω_x	-30-0-+30	Deg/s	0.1	1000	S	1
9.	Angular acceleration around lateral axis (ω_{y})	-60-0-+60	Deg/s	0.1	1000	S	1
10.	Angular acceleration around normal axis $(\omega_{z)}$	-10-0-+10	Deg/s	0.1	1000	S	1
11.	Dʒm (Displacement in Longitudinal) /Dʒe (Elevator displacement)	-30-0-+30	Deg	0.1	30	S	1
12.	Dʒn (Rudder displacement)	-30-0-+30	Deg	0.1	30	S	1

<u>Ser</u>	<u>Data</u>	<u>Limits</u>	<u>Unit</u>	<u>Least</u> Count	<u>Sampling</u> rate (Hz)	<u>Stand</u> <u>Alone (S) /</u> <u>Tapped (T)</u>	<u>No of</u> <u>Sensors</u> per Aircraft
13.	Dʒl (Displacement in lateral) /Dʒa (Aileron displacement)	-30-0-+30	Deg	0.1	30	S	1
14.	Dʒt (Displacement of throttle)	-5-0-+30	Deg	0.1	30	S	1
15. 16. 17.	Fuel Flow Rate per Engine q(FFR)	As Applicable	Ltr/Hrs Kgs/hrs Lb/Hrs	As Applicable	30	т	2
18.	Q (Torque per engine)	As Applicable	N-M	As Applicable	1000	S / T	2
19.	DGPS Time	As Applicable	S	As Applicable	1000	S	1
20.	DGPS Lat / Long	As Applicable	N/E	As Applicable	1000	S	1
21.	DGPS GS	As Applicable	Kt	As Applicable	1000	S	1

<u>Ser</u>	<u>Data</u>	<u>Limits</u>	<u>Unit</u>	<u>Least</u> Count	<u>Sampling</u> rate (Hz)	<u>Stand</u> <u>Alone (S) /</u> Tapped (T)	<u>No of</u> <u>Sensors</u> per Aircraft
22.	DGPS Track	As Applicable	Deg	As Applicable	1000	S	1
23.	TGT for each engine	As Applicable	deg C	As Applicable	30	Т	2
24.	Engine RPM for each engine (N_g)	As Applicable	Rpm	As Applicable	30	т	2
25.	Fuel Q _⊺	As Applicable	Kgs/Ltr	As Applicable	10	Т	1
26.	OAT	As Applicable	Deg C	As Applicable	10	Т	1
27.	Radio Altimeter Z _{RA}	As Applicable	Ft	As Applicable	10	Т	1
28.	Pressure Altitude Z _P	As Applicable	Ft	As Applicable	10	т	1
29.	Control Force in Longitudinal Axis F _M	-80-0-+80	DaN	0.1	30	S	1

<u>Ser</u>	<u>Data</u>	<u>Limits</u>	<u>Unit</u>	<u>Least</u> Count	<u>Sampling</u> rate (Hz)	<u>Stand</u> <u>Alone (S) /</u> <u>Tapped (T)</u>	<u>No of</u> <u>Sensors</u> per Aircraft
30.	Control Force in Lateral Axis F∟	-40-0-+40	DaN	0.1	30	S	1
31.	Rudder Control Force F_N	-80-0-+80	DaN	0.1	30	S	1
32.	Audio	50Hz-15KHz	As Applicable	As Applicable	As applicable (AA)	S	1
33.	Angle of attack	As Applicable	Deg	As Applicable	1000	S	1
34.	Side Slip angle	As Applicable	Deg	As Applicable	1000	S	1
35.	Tri axial accelerometer data for flutter analysis	As Applicable	G	As Applicable	1000	S	10
36.	Temperature compensated Strain gauges values for structural integrity tests *(Capability to measure force / stress / strain on arrester hook and under carriage system must	As Applicable	micro strains	As Applicable	1000	S	50

<u>Ser</u>	<u>Data</u>	<u>Limits</u>	<u>Unit</u>	<u>Least</u> Count	<u>Sampling</u> rate (Hz)	<u>Stand</u> <u>Alone (S) /</u> <u>Tapped (T)</u>	<u>No of</u> <u>Sensors</u> per Aircraft
	be provided						
37.	Total temperature	As Applicable	Deg C	As Applicable	10	S	1
38.	Absolute and differential pressures	As Applicable	Р	As Applicable	30	S	1
39.	Externally mounted High-speed camera	As Applicable	Video	As Applicable	1000	S	1
40.	Cockpit mounted High speed camera displays	As Applicable	Video	As Applicable	1000	S	1
41.	Externally mounted High-speed IR camera	As Applicable	Video	As Applicable	1000	S	1

<u>Ser</u>	<u>Data</u>	<u>Limits</u>	<u>Unit</u>	<u>Least</u> Count	<u>Sampling</u> rate (Hz)	<u>Stand</u> <u>Alone (S) /</u> <u>Tapped (T)</u>	<u>No of</u> <u>Sensors</u> per Aircraft
42.	Mil bus data	As Applicable	As Applicable	As Applicable	As Applicable	т	1
43.	ARINC bus data	As Applicable	As Applicable	As Applicable	As Applicable	т	1
44.	Wind True(Speed/Dir)	As Applicable	Kt/Deg	0.1/0.1	30	S	1
45.	Oleo displacement	As Applicable	ft	0.01	1000	S	3

HARDWARE (SENSORS) REQUIRED FOR PARAMETERS TO BE CAPTURED USING INSTRUMENTATION ON SHIP

<u>Ser</u>	Data	Limits	<u>Unit</u>	<u>Least</u> <u>Count</u>	<u>Sampling</u> <u>rate (Hz)</u>	<u>Stand</u> <u>Alone (S) /</u> Tapped (T)	<u>No of</u> <u>Sensors</u> per Ship
1.	Video	Up to 10 MP	MP	-	1000	S	1
2.	Roll	-40-0-+40	Deg	0.1	1000	S	1
3.	Pitch	-30-0-+30	Deg	0.1	1000	S	1
4.	Roll rate	Up to 30 deg / sec	Deg / sec	0.1	1000	S	1
5.	Pitch rate	Up to 30 deg / sec	Deg / sec	0.1	1000	S	1
6.	Yaw rate	Up to 30 deg / sec	Deg / sec	0.1	1000	S	1
7.	HDG	0-360	Deg	0.1	1000	S	1
8.	Heave	-6-0-+6	Мр	0.1	1000	S	1
9.	Heel	-10-0-+10	Deg	0.1	1000	S	1
10.	List	-10-0-+10	Deg	0.1	1000	S	1
11.	DGPS Time	As Applicable	S	As Applicable	1000	S	1
12.	DGPS Lat/Long	As Applicable	N/E	As Applicable	1000	S	1

<u>Ser</u>	<u>Data</u>	<u>Limits</u>	<u>Unit</u>	<u>Least</u> Count	<u>Sampling</u> rate (Hz)	<u>Stand</u> <u>Alone (S) /</u> <u>Tapped (T)</u>	<u>No of</u> <u>Sensors</u> per Ship
13.	DGPS GS	As Applicable	Kt	As Applicable	1000	S	1
14.	DGPS Track	As Applicable	Deg	As Applicable	1000	S	1
15.	Wind True(Speed/Dir)	As Applicable	Kt/Deg	0.1/0.1	30	S	1
16.	Wind Relative(Speed/Dir)	As Applicable	Kt/Deg	0.1/0.1	30	S	1
17.	Density Altitude	As Applicable	Ft	As Applicable	30	S	1
18.	OAT	As Applicable	Deg C	As Applicable	10	т	1
19.	Audio	50Hz-15KHz	As Applicable	As Applicable	As applicable (AA)	S	2
20.	Arresting Gear instrumentation for pressure recording	As Applicable	Ра	0.01	1000	S	6
21.	Arresting Gear instrumentation for cylinder displacement	As Applicable	ft	0.01	1000	S	6

<u>Ser</u>	<u>Data</u>	<u>Limits</u>	<u>Unit</u>	<u>Least</u> Count	<u>Sampling</u> rate (Hz)	<u>Stand</u> <u>Alone (S) /</u> <u>Tapped (T)</u>	<u>No of</u> <u>Sensors</u> per Ship
	recording						
22.	Arresting wire instrumentation for wire pull out / Displacement recording	As Applicable	ft	0.01	1000	S	3
23.	Arresting wire instrumentation for wire strain measurement (towards computation of wire loads)	As Applicable	Micro strains	1	1000	S	6
24.	Arresting wire instrumentation for wire strain measurement at mounting bolts (towards computation of bolt loads)	As Applicable	Micro strains	1	1000	S	6
25.	RGS instrumentation for pressure recording	As Applicable	Pa	0.01	1000	S	6
26.	RGS instrumentation for cylinder displacement recording	As Applicable	ft	0.01	1000	S	6
27.	Status of engines, stabilisers, ship's displacement, ship's trim (as text msg from a						1

<u>Ser</u>	<u>Data</u>	<u>Limits</u>	<u>Unit</u>	<u>Least</u> Count	<u>Sampling</u> rate (Hz)	<u>Stand</u> <u>Alone (S) /</u> <u>Tapped (T)</u>	<u>No of</u> <u>Sensors</u> per Ship
	console on ship to be displayed in aircraft)						

HARDWARE (SENSORS) REQUIRED FOR PARAMETERS TO BE CAPTURED USING INSTRUMENTATION ON HELICOPTER

<u>Ser</u>	<u>Data</u>	<u>Limits</u>	<u>Unit</u>	<u>Least</u> Count	<u>Sampling</u> rate (Hz)	<u>Stand</u> <u>Alone (S) /</u> <u>Tapped (T)</u>	<u>No of</u> Sensors
1.	Video	Up to 10 MP	MP	-	1000	Т	2
2.	Roll	-180-0-+180	Deg	0.1	1000	S	1
3.	Pitch	-90-0-+90	Deg	0.1	1000	S	1
4.	Yaw / Hdg	0-360	Deg	0.1	1000	S	1
5.	Linear acceleration along longitudinal axis (Nx)	3	G	0.1	1000	S	1
6.	Linear acceleration along lateral axis (Ny)	3	G	0.1	1000	S	1
7.	Linear acceleration along normal axis (Nz)	10	G	0.1	1000	S	1

<u>Ser</u>	<u>Data</u>	<u>Limits</u>	<u>Unit</u>	<u>Least</u> Count	<u>Sampling</u> rate (Hz)	<u>Stand</u> <u>Alone (S) /</u> <u>Tapped (T)</u>	<u>No of</u> Sensors
8.	Angular acceleration around longitudinal axis ω_x	-30-0-+30	Deg/s	0.1	1000	S	1
9.	Angular acceleration around lateral axis (ω_{y})	-60-0-+60	Deg/s	0.1	1000	S	1
10.	Angular acceleration around normal axis $(\omega_{z)}$	-10-0-+10	Deg/s	0.1	1000	S	1
11.	Dʒm (Displacement in Longitudinal)	-30-0-+30	Deg	0.1	30	S	1
12.	Dʒn (Rudder displacement)	-30-0-+30	Deg	0.1	30	S	1

<u>Ser</u>	<u>Data</u>	<u>Limits</u>	<u>Unit</u>	<u>Least</u> Count	<u>Sampling</u> rate (Hz)	<u>Stand</u> <u>Alone (S) /</u> <u>Tapped (T)</u>	<u>No of</u> Sensors
13.	Dʒl (Displacement in lateral)	-30-0-+30	Deg	0.1	30	S	1
14.	Dʒq (Collective displacement)	-5-0-+30	Deg	0.1	30	S	1
15. 16. 17.	Fuel Flow Rate per Engine q(FFR)	As Applicable	Ltr/Hrs Kgs/hrs Lb/Hrs	As Applicable	30	т	2
18.	Q (Torque per engine)	As Applicable	N-M	As Applicable	1000	S/T	2
19.	DGPS Time	As Applicable	S	As Applicable	1000	S	1
20.	DGPS Lat / Long	As Applicable	N/E	As Applicable	1000	S	1
21.	DGPS GS	As Applicable	Kt	As Applicable	1000	S	1

<u>Ser</u>	<u>Data</u>	<u>Limits</u>	<u>Unit</u>	<u>Least</u> Count	<u>Sampling</u> <u>rate (Hz)</u>	<u>Stand</u> <u>Alone (S) /</u> Tapped (T)	<u>No of</u> Sensors
22.	DGPS Track	As Applicable	Deg	As Applicable	1000	S	1
23.	Rotor RPM (N _R)	As Applicable	Rpm	As Applicable	30	т	1
24.	TGT for each engine	As Applicable	deg C	As Applicable	30	т	2
25.	Engine RPM for each engine (N_g)	As Applicable	Rpm	As Applicable	30	т	2
26.	Fuel Q _⊺	As Applicable	Kgs/Ltr	As Applicable	10	Т	1
27.	OAT	As Applicable	Deg C	As Applicable	10	т	1
28.	Radio Altimeter Z _{RA}	As Applicable	Ft	As Applicable	10	т	1
29.	Pressure Altitude Z_P	As Applicable	Ft	As Applicable	10	Т	1

<u>Ser</u>	<u>Data</u>	<u>Limits</u>	<u>Unit</u>	<u>Least</u> Count	<u>Sampling</u> rate (Hz)	<u>Stand</u> <u>Alone (S) /</u> Tapped (T)	<u>No of</u> Sensors
30.	Control Force in Longitudinal Axis F _M	-80-0-+80	DaN	0.1	30	S	1
31.	Control Force in Lateral Axis F∟	-40-0-+40	DaN	0.1	30	S	1
32.	Rudder Control Force F_N	-80-0-+80	DaN	0.1	30	S	1
33.	Audio	50Hz-15KHz	As Applicable	As Applicable	As applicable (AA)	S	1
34.	Angle of attack	As Applicable	Deg	As Applicable	1000	S	1
35.	Side Slip angle	As Applicable	Deg	As Applicable	1000	S	1
36.	Tri axial accelerometer data for flutter analysis	As Applicable	G	As Applicable	1000	S	10

<u>Ser</u>	<u>Data</u>	<u>Limits</u>	<u>Unit</u>	<u>Least</u> Count	<u>Sampling</u> rate (Hz)	<u>Stand</u> <u>Alone (S) /</u> Tapped (T)	<u>No of</u> Sensors
37.	Temperature compensated Strain gauges values for structural integrity tests for various locations	As Applicable	micro strains	As Applicable	1000	S	50
38.	Total temperature	As Applicable	Deg C	As Applicable	10	S	1
39.	Absolute and differential pressures	As Applicable	Р	As Applicable	30	S	1
40.	Externally mounted High-speed camera	As Applicable	Video	As Applicable	1000	S	1
41.	Cockpit mounted High speed camera displays	As Applicable	Video	As Applicable	1000	S	1
42.	Externally mounted High-speed IR camera	As Applicable	Video	As Applicable	1000	S	1

<u>Ser</u>	<u>Data</u>	<u>Limits</u>	<u>Unit</u>	<u>Least</u> Count	<u>Sampling</u> rate (Hz)	<u>Stand</u> <u>Alone (S) /</u> <u>Tapped (T)</u>	<u>No of</u> Sensors
43.	Mil bus data	As Applicable	As Applicable	As Applicable	As Applicable	Т	1
44.	ARINC bus data	As Applicable	As Applicable	As Applicable	As Applicable	Т	1
45.	Wind True(Speed/Dir)	As Applicable	Kt/Deg	0.1/0.1	30	S	1
46.	Oleo displacement	As Applicable	ft	0.01	1000	S	3

HARDWARE (SENSORS) REQUIRED FOR PARAMETERS TO BE CAPTURED USING INSTRUMENTATION ON MOBILE TELEMETRY

<u>Ser</u>	<u>Data</u>	<u>Limits</u>	<u>Unit</u>	<u>Least</u> Count	<u>Sampling</u> <u>rate (Hz)</u>	<u>Stand</u> <u>Alone (S) /</u> <u>Tapped (T)</u>	<u>No of</u> <u>Sensors</u> <u>per mobile</u> <u>telemetry</u>
1.	DGPS Time	As Applicable	S	As Applicable	>10	S	1
2.	DGPS Lat/Long	As Applicable	N/E	As Applicable	>10	S	1
3.	DGPS GS	As Applicable	Kt	As Applicable	>10	S	1
4.	DGPS Track	As Applicable	Deg	As Applicable	>10	S	1
5.	Wind True(Speed/Dir)	As Applicable	Kt/Deg	0.1/0.1	>2	S	1
6.	Wind Relative(Speed/Dir)	As Applicable	Kt/Deg	0.1/0.1	>2	S	1
7.	Density Altitude	As Applicable	Ft	As Applicable	>1	S	1

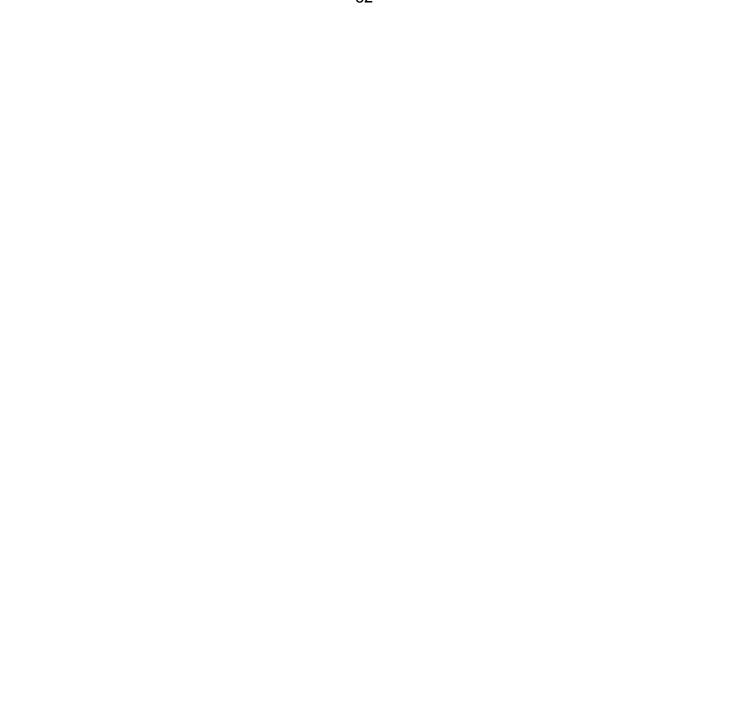
<u>Ser</u>	<u>Data</u>	<u>Limits</u>	<u>Unit</u>	<u>Least</u> Count	<u>Sampling</u> rate (Hz)	<u>Stand</u> <u>Alone (S) /</u> <u>Tapped (T)</u>	<u>No of</u> <u>Sensors</u> per mobile <u>telemetry</u>
8.	Pressure Altitude	As Applicable	Ft	As Applicable	>1	S	1
9.	OAT	As Applicable	Deg C	As Applicable	>1	т	1
10.	Video (the video camera must be able to capture / track the aircraft when in visual range)	Up to 10 MP	MP		30	т	1
11.	Pressure Error Correction (PEC) trials equipments encompassing :- (a) Provision of TV tracking system (b) Precision barometer (c) Thermometer (d) Laser based vehicle position and speed / direction measurement equipment (e) High Speed Camera			As per COTS			1 each

<u>Ser</u>	<u>Data</u>	<u>Limits</u>	<u>Unit</u>	<u>Least</u> Count	<u>Sampling</u> rate (Hz)	<u>Stand</u> <u>Alone (S) /</u> <u>Tapped (T)</u>	<u>No of</u> <u>Sensors</u> <u>per mobile</u> <u>telemetry</u>
12.	 Analsyis provision and facilities :- (a) Work bench (b) Oscilloscope (c) Spectrum Analyser (d) Provision for two- way audio and video interface with aircraft (e) Associated monitoring consoles for minimum three receivers 			As per COTS			1 each

HARDWARE (SENSORS) REQUIRED FOR PARAMETERS TO BE CAPTURED USING INSTRUMENTATION ON SHIP

<u>Ser</u>	Data	Limits	Unit	<u>Least</u> <u>Count</u>	<u>Sampling</u> <u>rate (Hz)</u>	<u>Stand</u> <u>Alone (S) /</u> Tapped (T)	<u>No of</u> Sensors per Ship
1.	Video	Up to 10 MP	MP	-	1000	S	1
2.	Roll	-40-0-+40	Deg	0.1	1000	S	1
3.	Pitch	-30-0-+30	Deg	0.1	1000	S	1
4.	Roll rate	Up to 30 deg / sec	Deg / sec	0.1	1000	S	1
5.	Pitch rate	Up to 30 deg / sec	Deg / sec	0.1	1000	S	1
6.	Yaw rate	Up to 30 deg / sec	Deg / sec	0.1	1000	S	1
7.	HDG	0-360	Deg	0.1	1000	S	1
8.	Heave	-6-0-+6	Мр	0.1	1000	S	1
9.	Heel	-10-0-+10	Deg	0.1	1000	S	1
10.	List	-10-0-+10	Deg	0.1	1000	S	1
11.	DGPS Time	As Applicable	S	As Applicable	1000	S	1
12.	DGPS Lat/Long	As Applicable	N/E	As Applicable	1000	S	1

<u>Ser</u>	<u>Data</u>	<u>Limits</u>	<u>Unit</u>	<u>Least</u> Count	<u>Sampling</u> rate (Hz)	<u>Stand</u> <u>Alone (S) /</u> <u>Tapped (T)</u>	<u>No of</u> <u>Sensors</u> per Ship
13.	DGPS GS	As Applicable	Kt	As Applicable	1000	S	1
14.	DGPS Track	As Applicable	Deg	As Applicable	1000	S	1
15.	Wind True(Speed/Dir)	As Applicable	Kt/Deg	0.1/0.1	30	S	1
16.	Wind Relative(Speed/Dir)	As Applicable	Kt/Deg	0.1/0.1	30	S	1
17.	Density Altitude	As Applicable	Ft	As Applicable	30	S	1
18.	OAT	As Applicable	Deg C	As Applicable	10	т	1
19.	Audio	50Hz-15KHz	As Applicable	As Applicable	As applicable (AA)	S	2
20.	Status of engines, stabilisers, ship's displacement, ship's trim (as text msg from a console on ship to be displayed in aircraft)			1		,	1



QUESTIONNAIRE FOR DETAILS IN RESPECT OF NAVAL AIRCRAFT SYSTEM INSTRUMENTATION AND TELEMETRY (NASI)

<u>Ser</u>	Information Required	Data / information to be provided by Vendor
	Vendor Qualification	_
1.	The vendor is to indicate details of Instrumentation	
	and telemetry tasks undertaken so far. The vendor is	
	to specify details of data captured and analysed for	
	instrumentation tasks undertaken.	
2.	What kind of trials has the firm been involved with its	
	instrumentation and telemetry facilities?	
3.	Has the firm been involved in preparation of	
	instrumentation and telemetry scheme? Quote the	
	requirements in generation of instrumentation	
	scheme.	
4.	Has the firm been associated with CEMILAC for	
	approval of instrumentation scheme? Quote the tasks	
	involved.	
5.	Is the vendor an OEM of Data Acquisition Unit?	
6.	What Indigenization Content IC is considered	
	implementable?	
7.	Foreign vendor to indicate the percentage of ToT	
	willing to execute with his Indian partner over make	
	portion of contract.	
8.	What is the experience of the firm on	
	manufacture/supply/ integration of Data Acquisition	
	System, displays with customised monitoring for	
	parameters and event marking provision.	
	Instrumentation and Telemetry.	
9.	Elaborate activities envisaged on aircraft, ships and	
	ground station for setting up of NASI to meet	
10	instrumentation and telemetry requirements.	
10.	What tools and test equipments would be required as	
11	part of instrumentation and telemetry.	
11.	Vendor to list down infrastructure augmentation at HL	
12.	and TL for setting up of NASI. Would the vendor be willing to provide compatible	
12.		
	interfaces for installation of sensors and equipment	
	for all ships and aircraft platforms.	

13.	Would the vendor ensure that all equipment used are	
	military grade?	
	Parameters.	
14.	Vendor to assess adequacy of parameters listed at	
	Appendices A to E and bring out additions if any.	
15.	The vendor is to submit technical specification of	
	sensors and equipment to fulfill the testing	
	requirements.	
16.	The vendor is to indicate the military specification	
	compliance / air worthiness compliance standards of	
	the equipment needed.	
17.	The vendor to provide a proposed instrumentation	
	scheme / layout using all sensors and equipment.	
18.	MTBF and MTTR of each equipment and sensors to	
	be indicated.	
	Mobile Telemetry Platform.	
19.	What all features would be provided on mobile	
	telemetry to facilitate Pressure Error Correction ⁹	
	Trials.	
20.	Which all equipment would be integrated in Mobile	
	telemetry to meet flight testing requirements as	
	described at para 5 (b) (i) (ab).	
21.	What audio and video interface would be provided in	
	mobile telemetry?	
22.	How many monitoring consoles would be provided in	
	mobile telemetry?	
23.	List out specifications for environmental control and	
0.1	power supply requirements in mobile telemetry.	
24.	What would be the specifications of rugged vehicle (in	
	terms of tonnage, speed, etc) which could facilitate	
	mobile telemetry requirements?	
25	Data Handling Module.	
25.	Which all feature would be provided in Data Handling	
	Module to achieve prediction capability?	
	What data analytics capability would be unsertaken in	
00	data handling module?	
26.	What would be the data storage capacity to cater for	
07	3 years of operation assuming 10 trials per month?	
27.	Real time processing capability needed for structural	

⁹ Pressure Error Correction (PEC) evaluation is undertaken to calibrate pressure instruments. This trial is needed to be performed when the airflow around the Pitot-Static probes are likely to get altered due to any change in external configuration.

	testing. What features can be provided in Data	
	Handling Module for the same?	
28.	Which data simulation facility would be provided for	
	such data handling?	
	Aircraft Weighing Module.	
29.	Is there a provision for automatic calculation of	
	vertical, lateral and longitudinal center of gravity?	
30.	What is the realistic life cycle for aircraft weighting	
	module.	
31.	The vendor would be required to undertake leveling	
	of the tarmac to install weighing provision. Would the	
	vendor be willing to undertake the task?	
32.	Vendor is required to list down scope of work and	
	components of aircraft weighing module.	
	Operation and Maintenance Module.	
33.	What are the tasks involved in meeting scope of work	
	listed for operation and maintenance module as	
	enumerated in Para 5 (e)?	
34.	What kind of specialization is desired to meet the	
	above scope of work?	
35.	How many specialist with what experience and	
	qualification would be suitable for the onsite	
	operational support?	
36.	What timelines are envisaged for each of the listed	
27	scope of work?	
37.	Would the vendor be willing to provide door to door	
38.	spare support?	
30.	What tools, test equipments are required for	
39.	operational and maintenance support? Would the vendor be willing to provide onsite support	
39.	of minimum six crew for operational and maintenance	
	· ·	
40.	support? Would the vendor be able to ensure continuity in	
40.	manpower?	
41.	What are the equipment/ tools and facilities needed	
	towards op and maintenance support?	
	Financial Aspects.	
42.	The vendor is to provide indicative cost for setting up	
	of NASI for each modules mentioned and any other	
	setup required to fulfill the envisaged role.	
43.	Breakdown of cost (separate for Indian and imported	
	components) is to be indicated.	

44.	The vendor should undertake installation and	
	commissioning of all equipment including associated	
	labs, setting up and facilities within three (03) months	
	of signing of contract. The vendor(s) is required to	
	indicate the feasibility of undertaking the same with	
	cost.	
	Documentation.	
45.	The vendor is required to provide detailed	
	documentation of all the equipment provided towards	
	Project NASI in hard and soft copy. Would the vendor	
	be willing to do the same?	
46.	The vendor is to indicate details of documentation	
	that are required to be sought as part of setting up of	
	NASI.	
	On Job Training.	
47.	During the course of setting up of NASI and rendering	
	operational and maintenance support, the vendor	
	would be required to impart on job training to naval	
	technical crew on the entire NASI set up. Would the	
	vendor be willing to undertake on job training?	
48.	What is the scope of training envisaged?	
	Additional Details.	
49.	The vendors may include any other additional details,	
	as deemed appropriate, to elaborate the proposal.	
50.	Vendor should confirm if following conditions would	
	be acceptable:-	
	(a) The solicitation of offers will be as per 'Single	
	Stage – Two Bid System'. It would imply that a	
	'Request for Proposal (RFP)' would be issued	
	soliciting technical and commercial offers together,	
	but in two separate sealed envelopes. The validity	
	of commercial offers should be at least 18 months	
	from the date of submission of offers.	
	(b) The technical offers would be evaluated to	
	check its compliance with RFP.	
	(c) Vendors qualifying the technical requirements	
	for the RFP would be shortlisted for evaluation of	
	commercial offers.	
	(d) Vendors once found suitable would be	
	shortlisted. A Contract Negotiation Committee	
	(CNC) would decide the lowest cost bidder (L1)	

and conclude the appropriate contract.	

INFORMATION PROFORMA

1. Name, Address and Unique ID (if any) of the Vendor /Company /Vendor.

(Company profile, in brief, to be attached. In the eventuality of the vendor emerging as L1, contract will be concluded in the name and address of the vendor, as indicated here). Vendors are to submit an undertaking that any subsequent proposal for change in name of vendor or address, will be intimated to IHQ MoD (Navy) at the first available opportunity and supporting documents be furnished within five working days of approval by relevant competent authority.

2. Type (Tick the relevant category).

Vendor manufacturing Telemetric and Instrumentation System (OEM) **Yes/No** Government sponsored Export Agency **Yes/No** (Details of registration to be provided) Authorised Vendor of Vendor manufacturing Telemetric and Instrumentation System **Yes/No** (attach details)

Others (give specific details)

3. Contact Details.

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Postal Address:		
City:	Province:	
Country:	Pin/Zip Code:	
Tele:	Fax:	

4. Local Branch/Liaison Office/ Authorised Representative, in India (if any).

Name & Postal Address:

City:	Province:	
Country:	Pin/Zip Code:	
Tele:	Fax:	
URL/Web Site:		

5. **Financial Details.**

Annual Turnover: ______ USD

Number of Employees in vendor

Details of manufacturing infrastructure available

Earlier contracts with Indian Ministry of Defence/ Government agencies:

(Agency Contract Number Equipment Quantity Cost)

6. Certification Quality Assurance Organisation (If Applicable). Name of Agency_____

7. Equipment/Product Profile (to be submitted for each product separately):-

(a) Name of Product:

(Should be given category wise for e.g. all products under night vision devices to be mentioned together)

(b) Description (attach technical literature):

(c) Whether OEM or Integrator: _____

(d) Status (in service /design & development stage): _____

(e) Production capacity per annum: _____

(f) Countries where equipment is in service:

(g) Whether export clearance is required from respective Government: _____

(h) Any collaboration/joint venture/ co production/ authorized dealer with foreign Industry (give details):

Name & Address: ______ Tel: Fax:

(j) Any other relevant information.

8. Declaration. It is certified that: -

(a) The above information is true and any changes will be intimated within five(05) working days of occurrence.

(b) The _____ (name of vendor) has never been banned/de-barred for doing business with MoD/Gol/any other government organisation and that there is no inquiry going on by CBI/ED/ any other government agency against the vendor.

(Authorised Signatory)